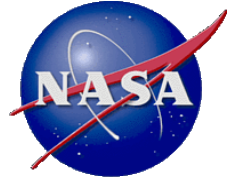


National Aeronautics and Space Administration



Development of the Space Debris Sensor (SDS)

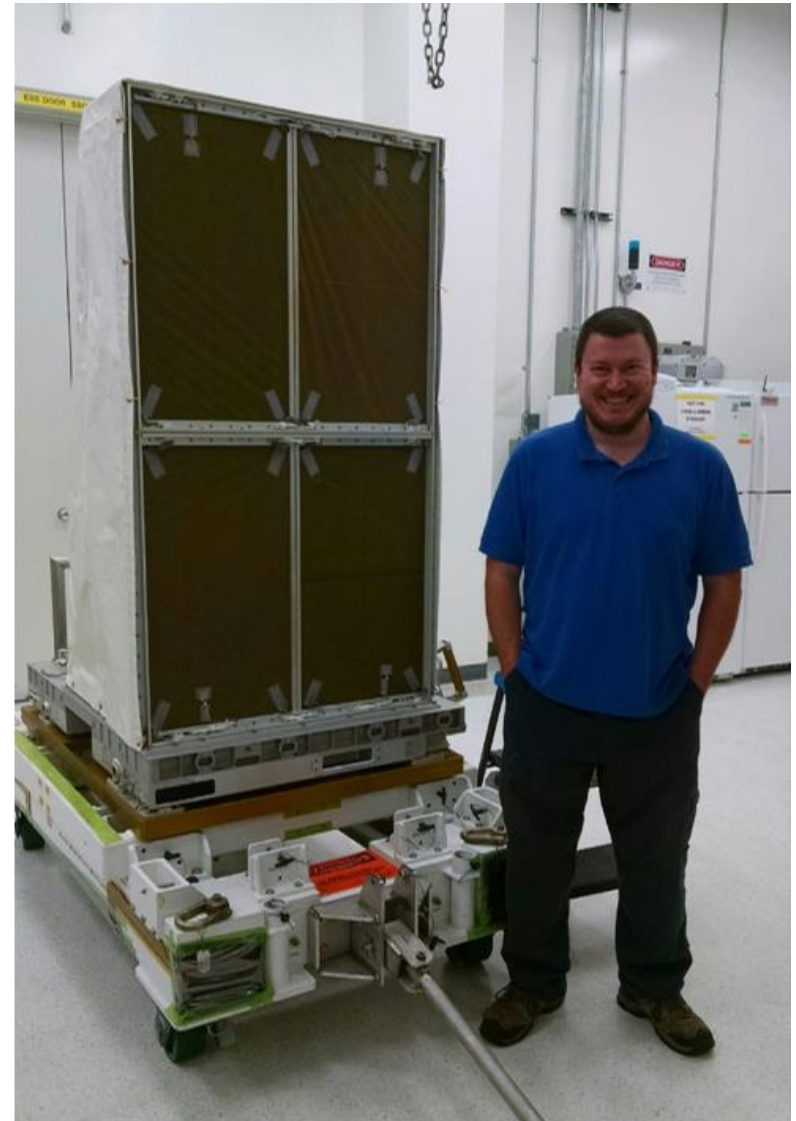


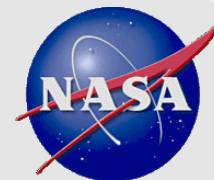
Joe Hamilton
SDS Principal Investigator
January 31. 2017



Outline

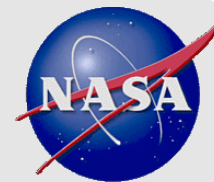
- **Background**
- **Orbital Debris Measurement Coverage**
- **What is SDS?**
- **Detection Principles**
- **Example 0.4 mm 30° Stainless Steel 7 km/s**
- **500µm 440C Stainless Steel**
- **500µm Aluminum Al 2017-T4**
- **500µm PMMA Plexiglass**
- **SDS Concept of Operations**
- **SDS on Columbus-External Payload Facility**
- **SDS ISS Orientation**
- **2-D Directional Flux – ORDEM 3.0**
- **2-D Directional Flux – MEM 2.0**
- **Predicted Flux vs. Velocity**
- **Conclusions**
- **Questions?**



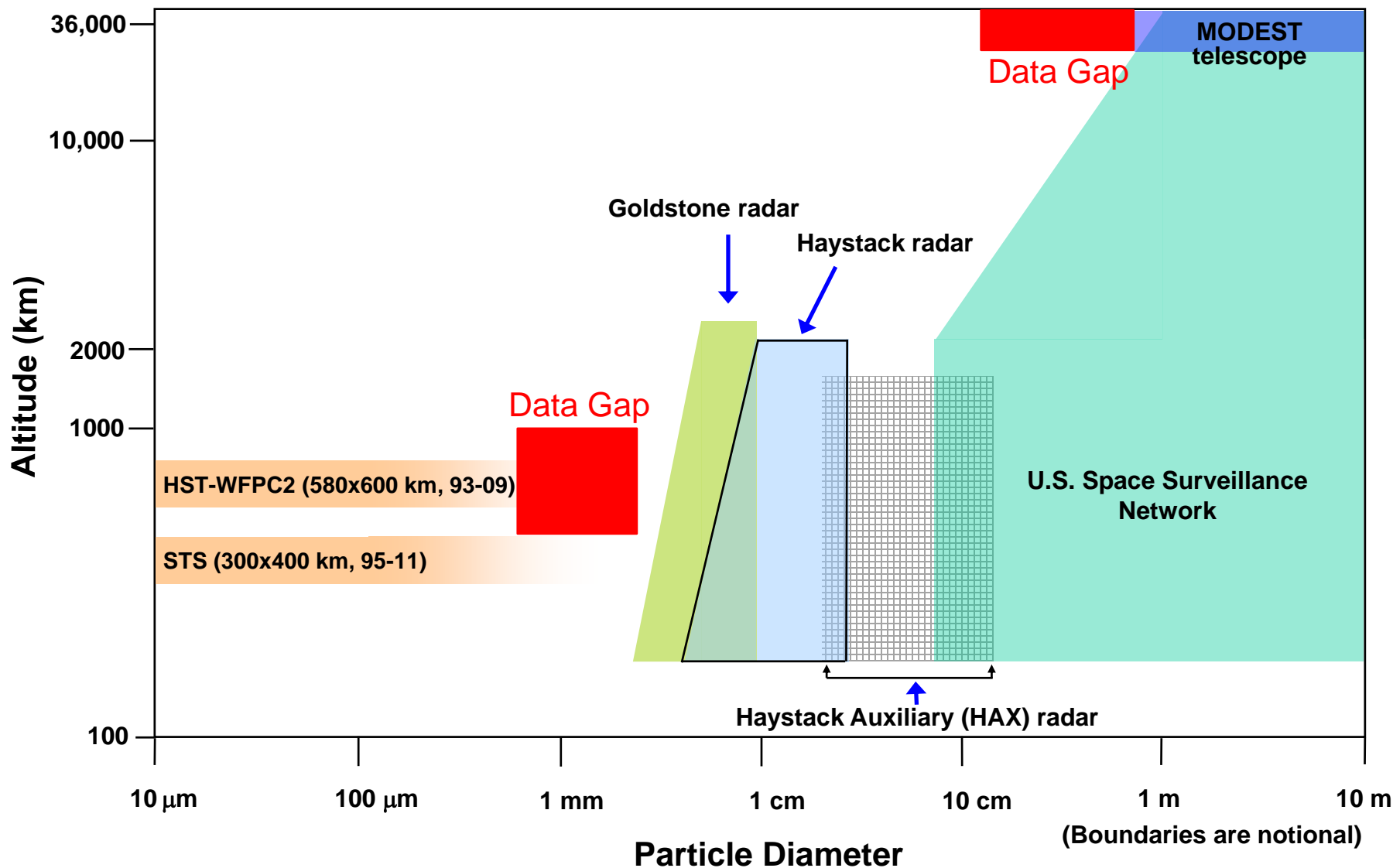


Background

- DRAGONS concept and technology has been under development with intermittent grants since 2002
- The goal of DRAGONS is to provide in-situ statistical data on the debris population that is too small for ground-based remote sensing to accomplish.
 - Results would be used to update the Orbital Debris Engineering Model (ORDEM)
 - **Current estimate of the small debris population is based on inspection of exposed surfaces returned on Shuttle (Retired 2011)**
- The DRAGONS team includes the NASA Orbital Debris Program Office, the NASA Hypervelocity Impact Technology group, the NASA/JSC Engineering Directorate, Jacobs, the United States Naval Academy, the Naval Research Lab, Virginia Tech, and the University of Kent.



Orbital Debris Measurement Coverage





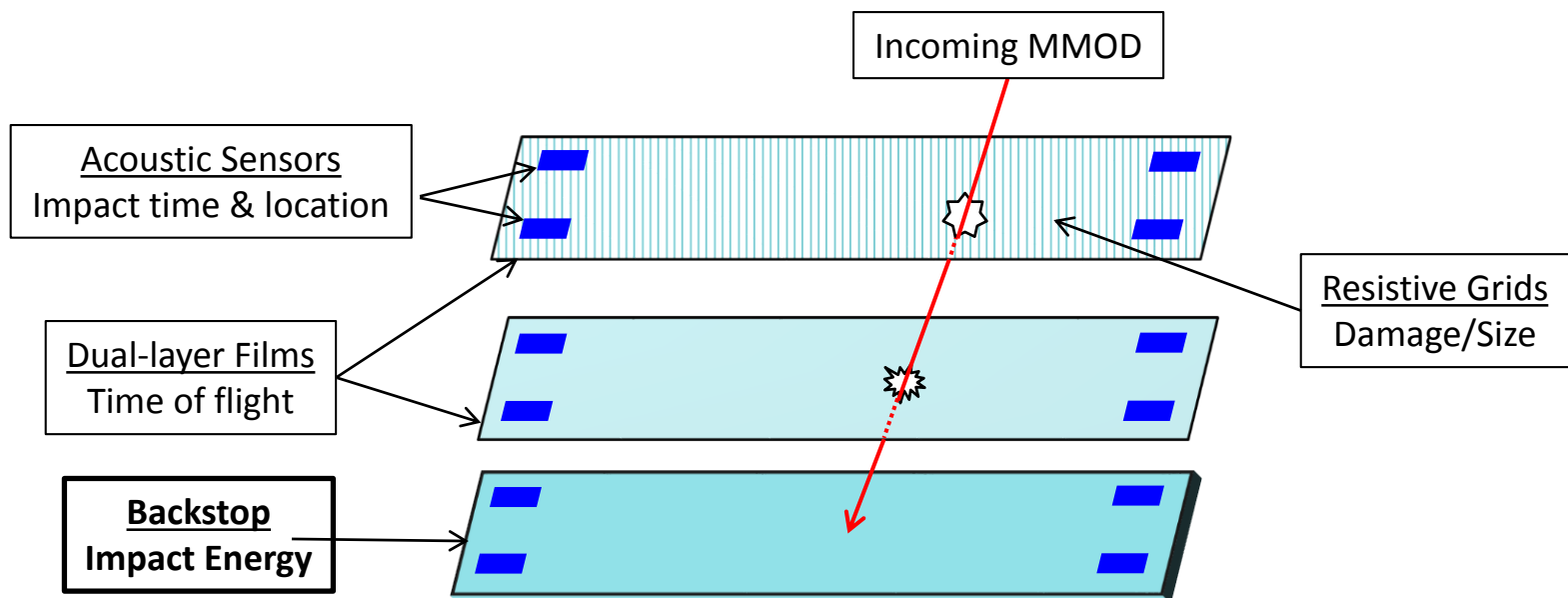
What is SDS?

- **DRAGONS is an impact sensor designed to detect and characterize collisions with small orbital debris.**
 - 50 μ m to > 1mm debris size detection
 - Characterize debris size, speed, direction, and density
- **The Space Debris Sensor (SDS) is a flight demonstration of DRAGONS on the International Space Station**
 - Approximately 1 m² of detection area facing the ISS velocity vector
 - Minimum two year mission on Columbus External Payloads Facility (EPF)
 - Minimal obstruction from ISS hardware
 - Development is nearing final checkout and integration with the ISS
 - Current launch schedule is SpaceX 13, ~ Sept 2017, or SpaceX 14, ~ Jan 2018

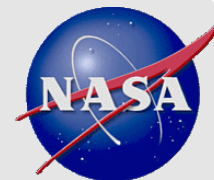


Detection Principles

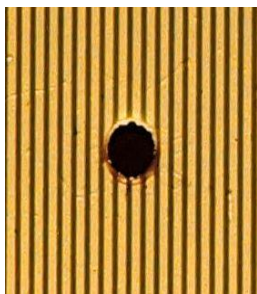
- SDS combines dual-layer thin films, an acoustic sensor system, a resistive grid sensor system, and sensed backstop
- Impact detection and recording capability
 - **Impact time, particle size, impact speed, impact direction, and impact energy/particle density**



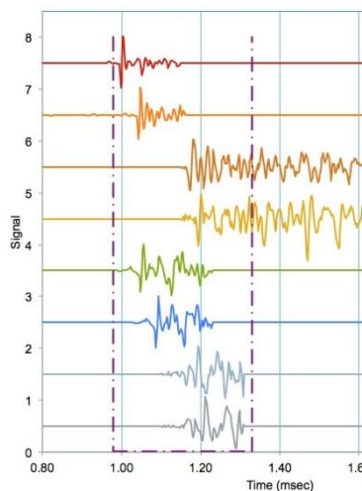
Example 0.4 mm 30° Stainless Steel 7 km/s



Layer 1 hole
Broke 3 lines

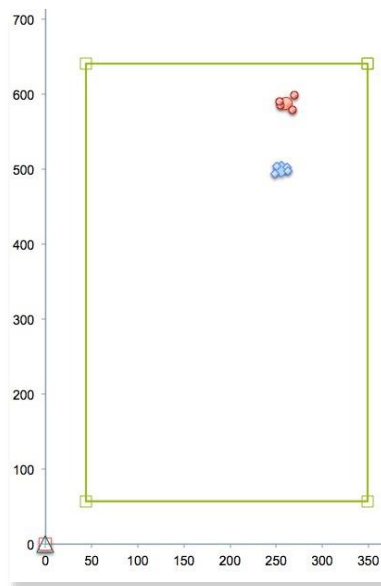


Filtered Acoustic Data

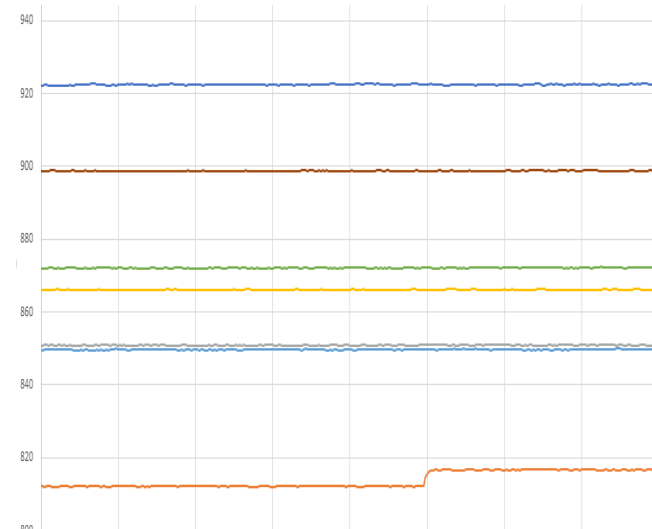


Location

Red=Layer 1;
Blue = Layer 2

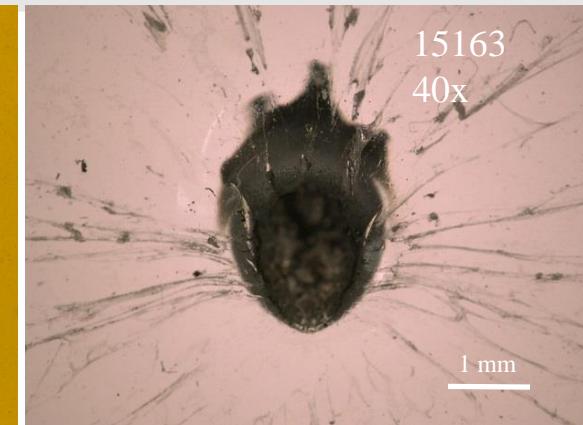
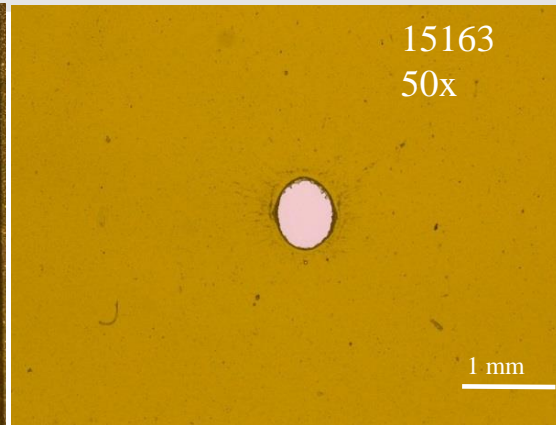
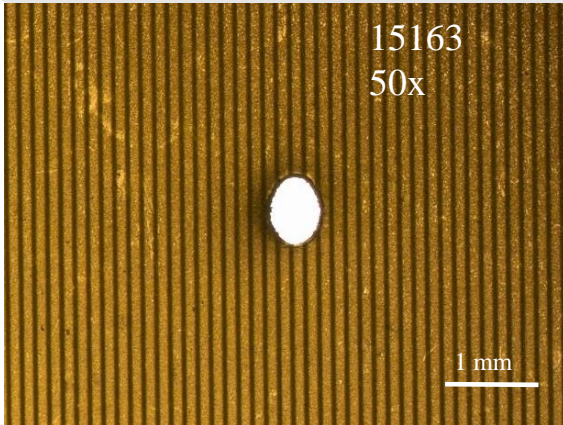


Resistance change
consistent with 3
line break

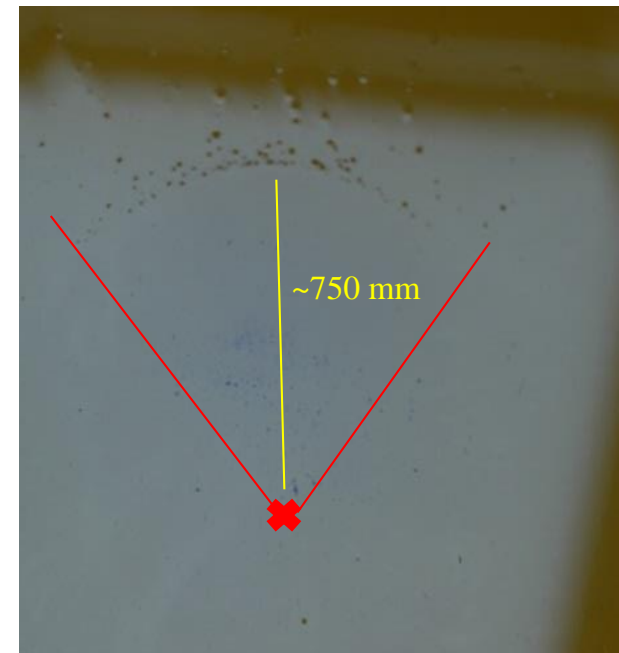




500 μ m 440C Stainless Steel

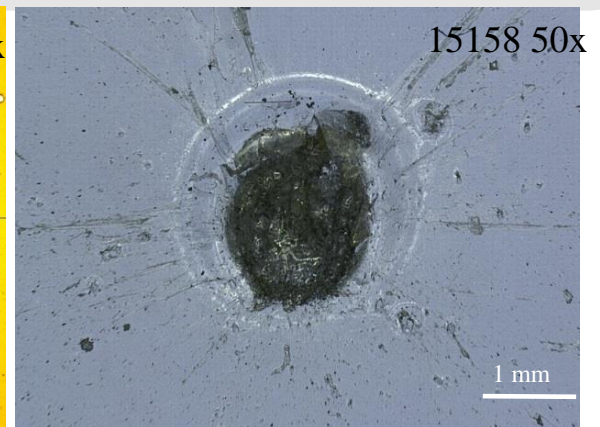
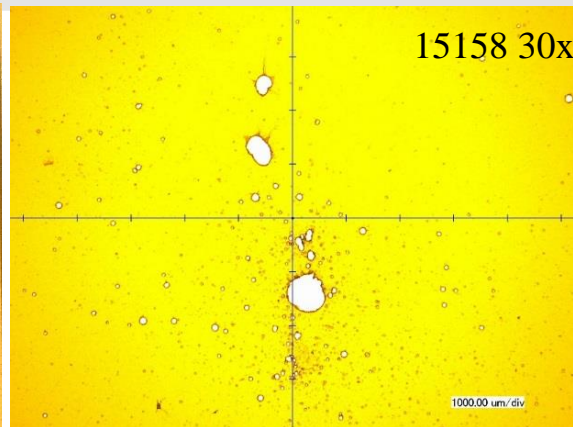
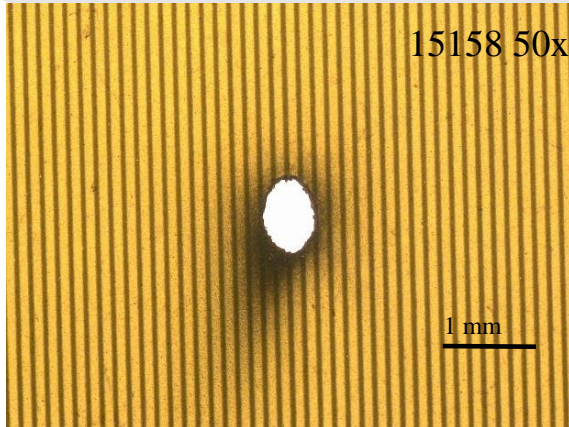


- Steel maintains shape throughout
- No visible break up of particles during impacts
- Steel shots produce significant secondary ejecta from Lexan back plate
- Ejecta has enough velocity to penetrate and dimple Kapton layer in wide arc downstream from shot
- Straight-on shots produce halo around entry hole; As shot angle increases the damage moves further away from hole

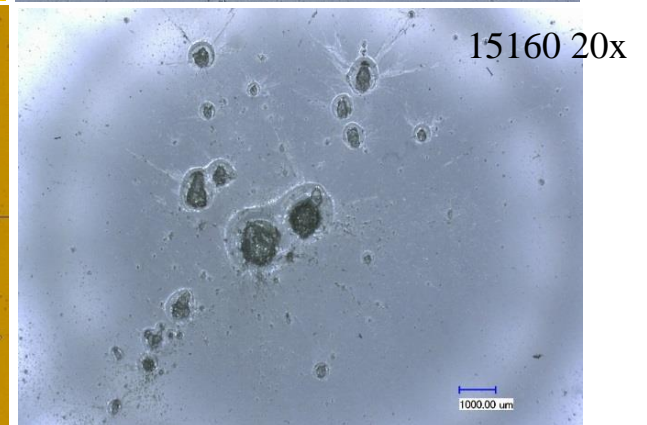
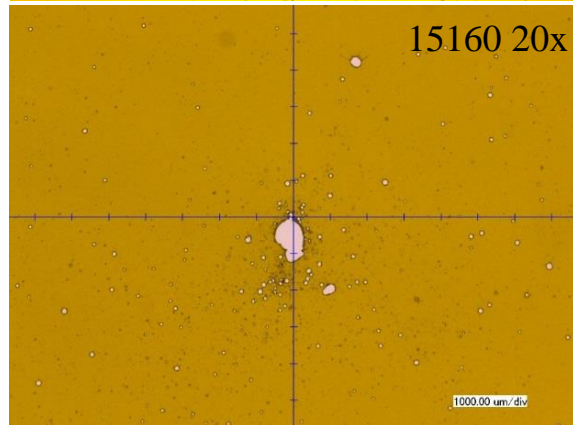


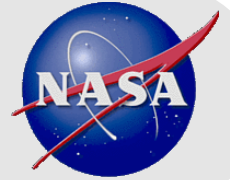


500 μ m Aluminum Al 2017-T4

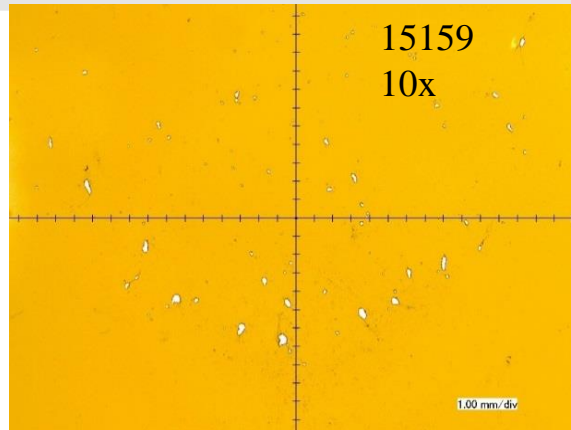
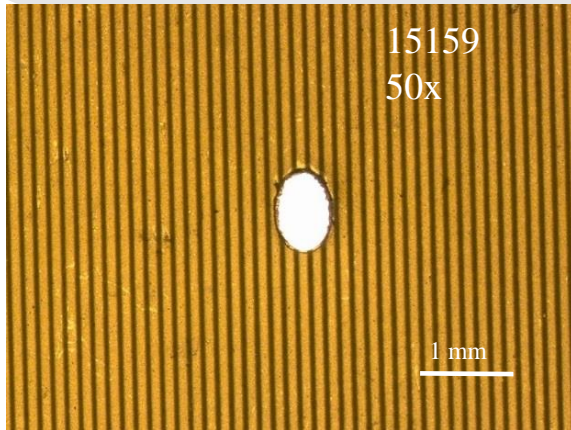


- Aluminum particles show break-up after first layer
- Amount of break up varied in the three shots
- One shot left a clean crater on Lexan back plate
- Two other shots had a collection of smaller craters
- No sign of ejecta damage on Kapton layer

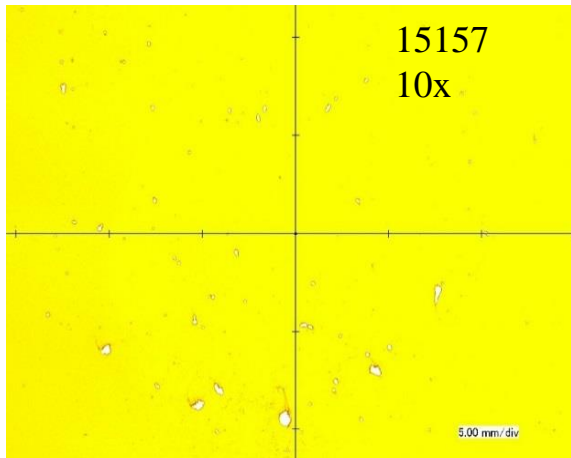




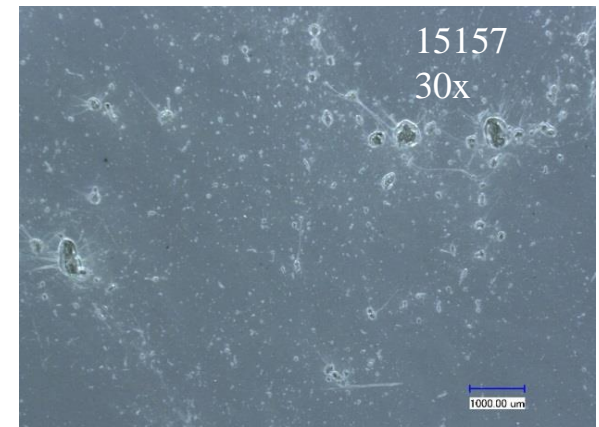
500 μ m PMMA Plexiglass

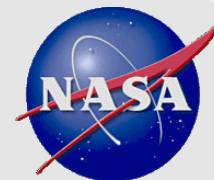


- Plastic particles broke up significantly after impacting the NCAS grid
- ‘Half circle’ hole pattern on Kapton layer with largest hole at bottom of the circle
- Same break up pattern for all shots - ~25mm wide by 20mm tall area of holes

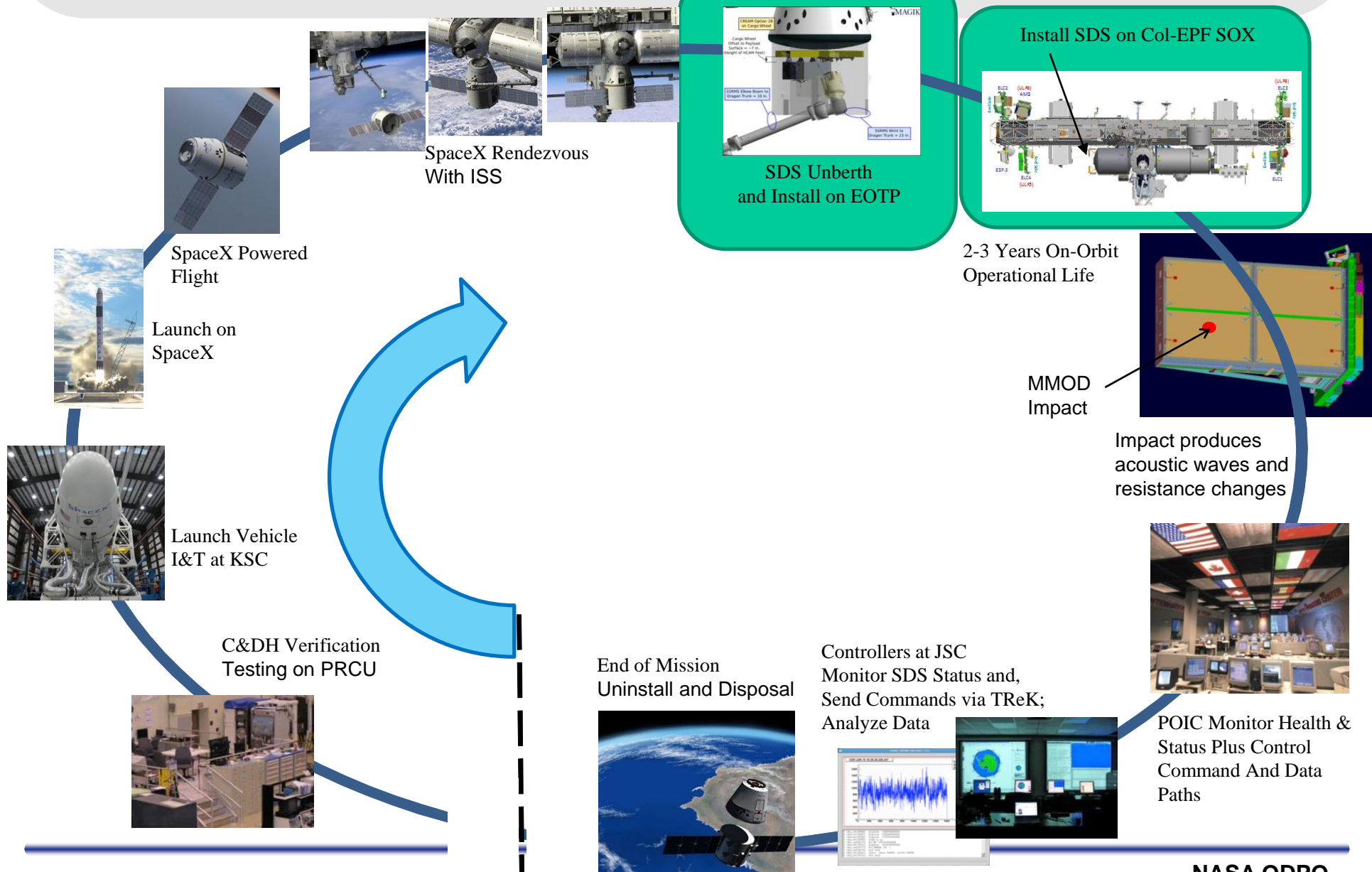


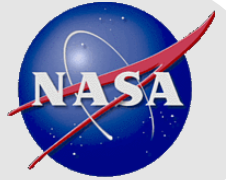
- Only one plastic shot showed up on the Lexan back plate
- No craters on Lexan – only residue



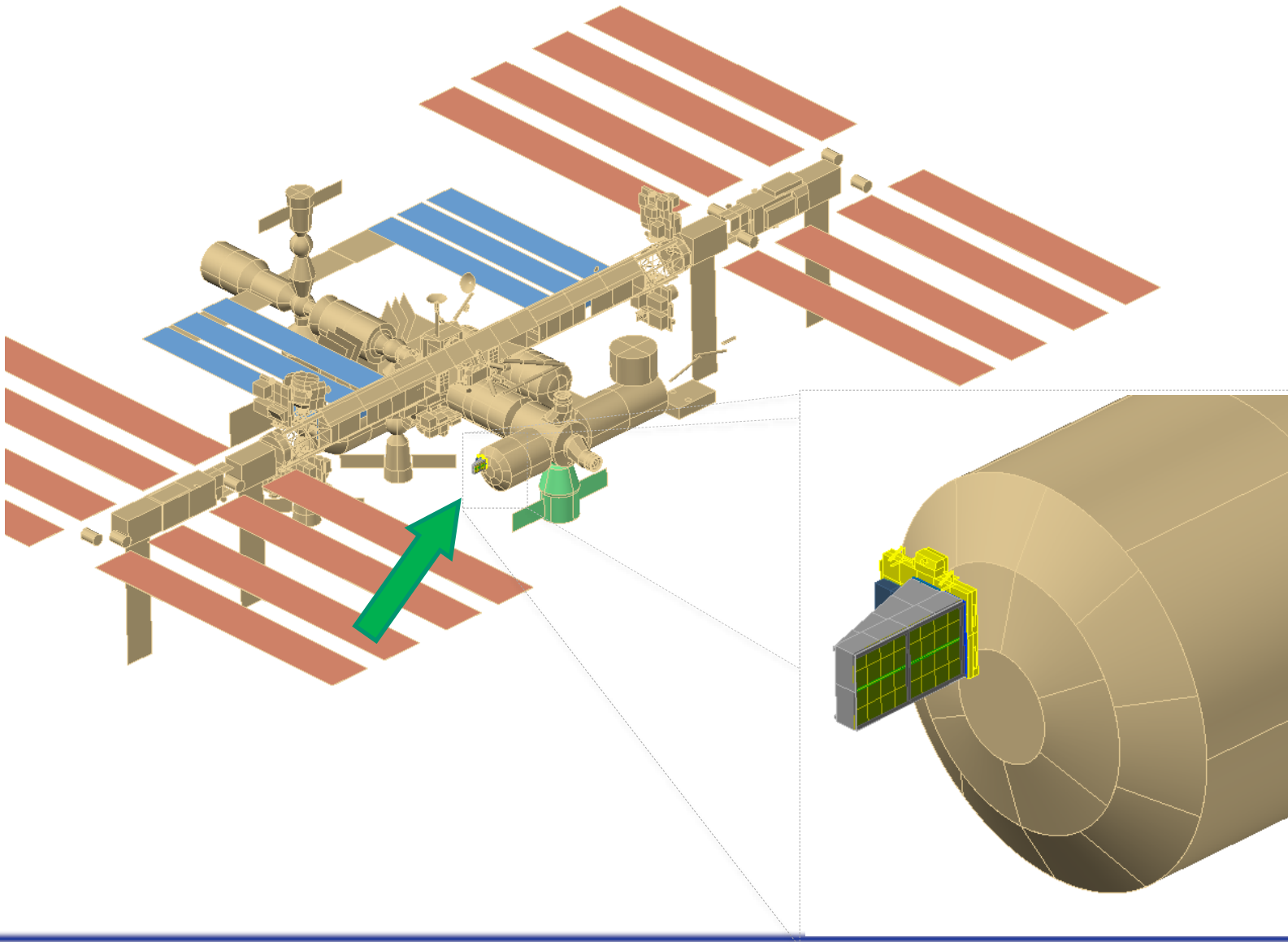


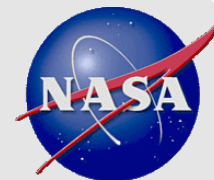
SDS Concept of Operations



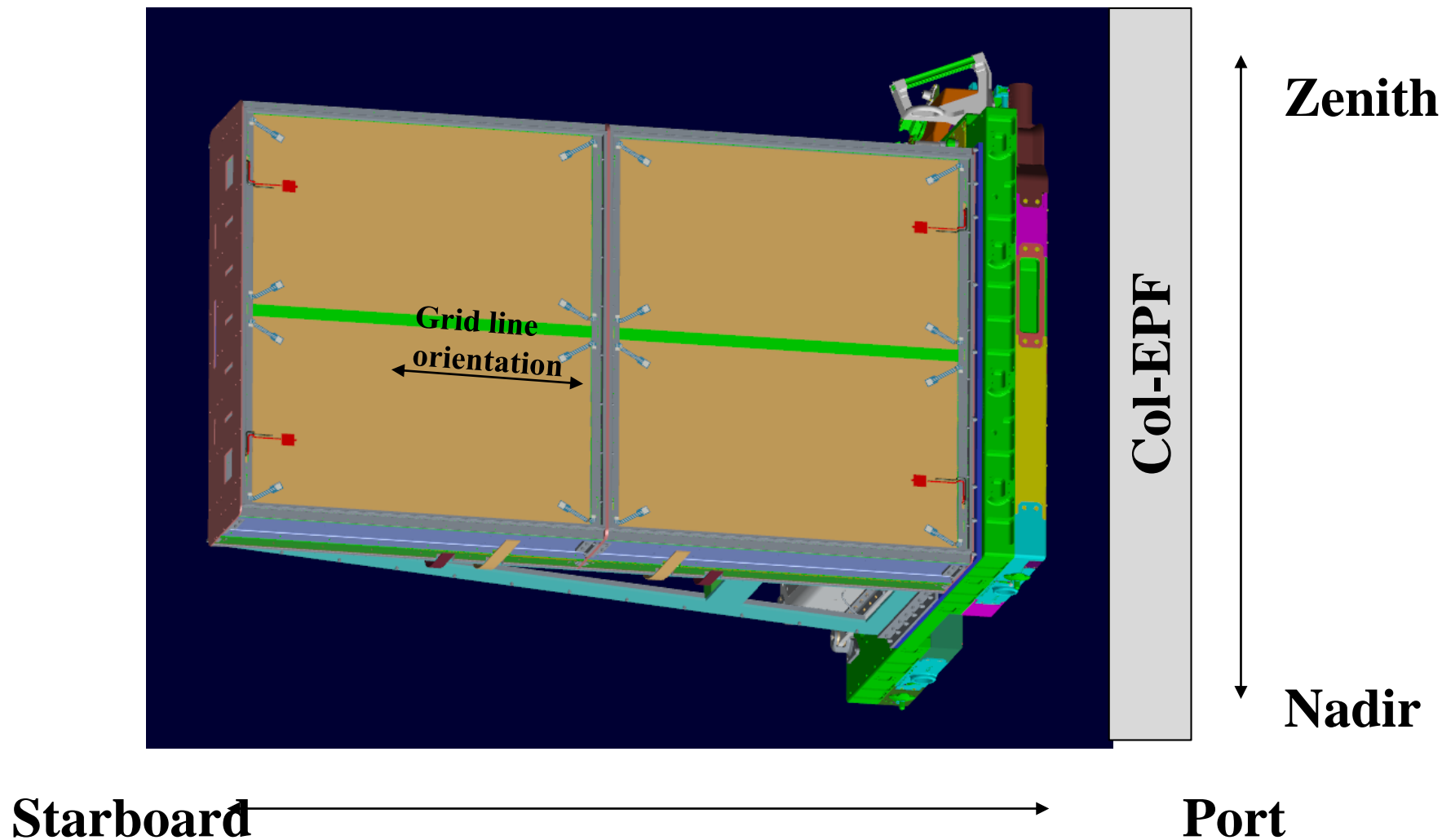


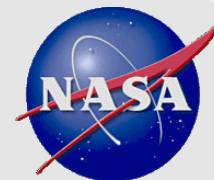
SDS on Columbus-External Payload Facility





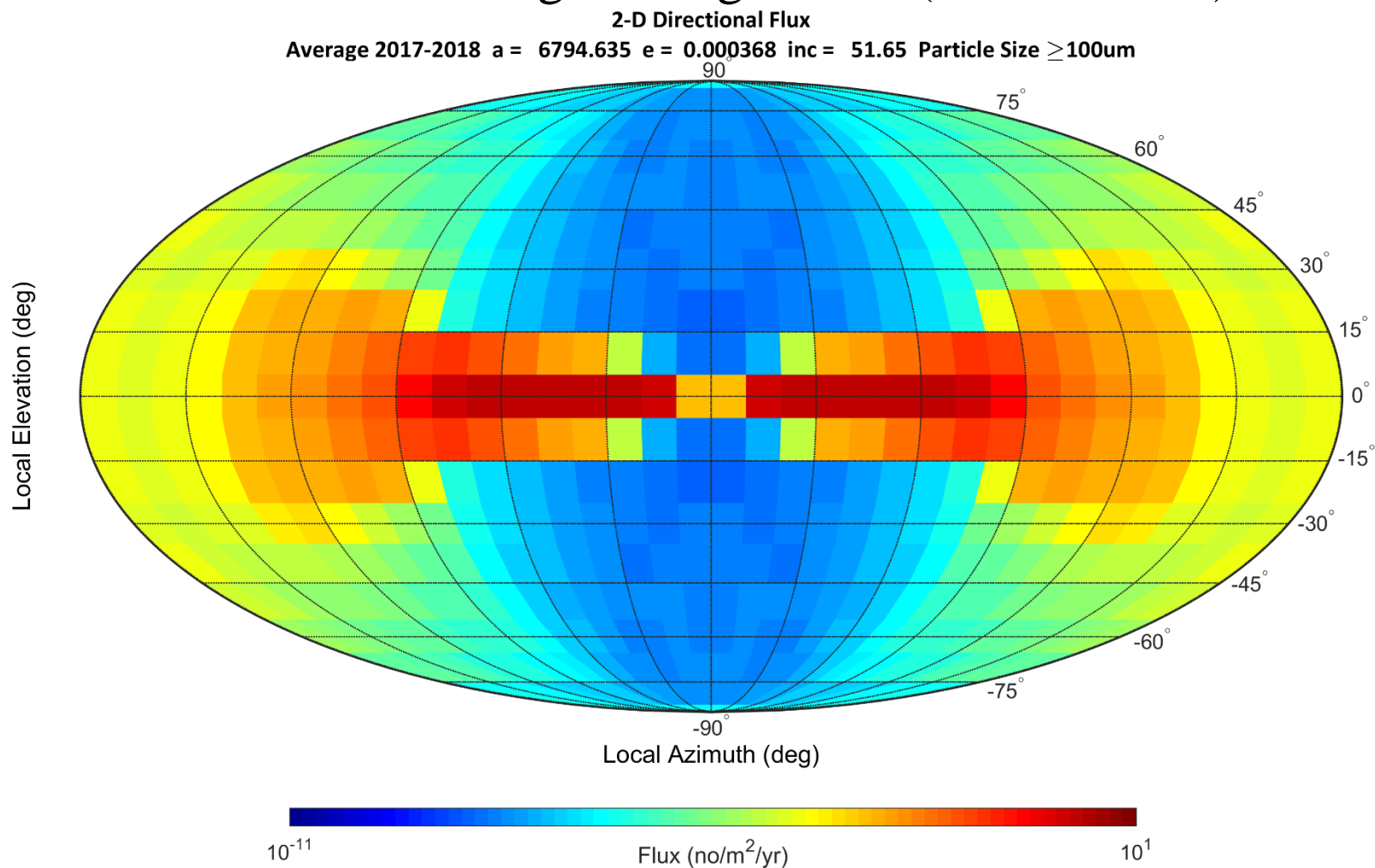
SDS ISS Orientation

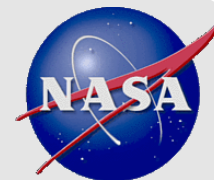




2-D Directional Flux – ORDEM 3.0

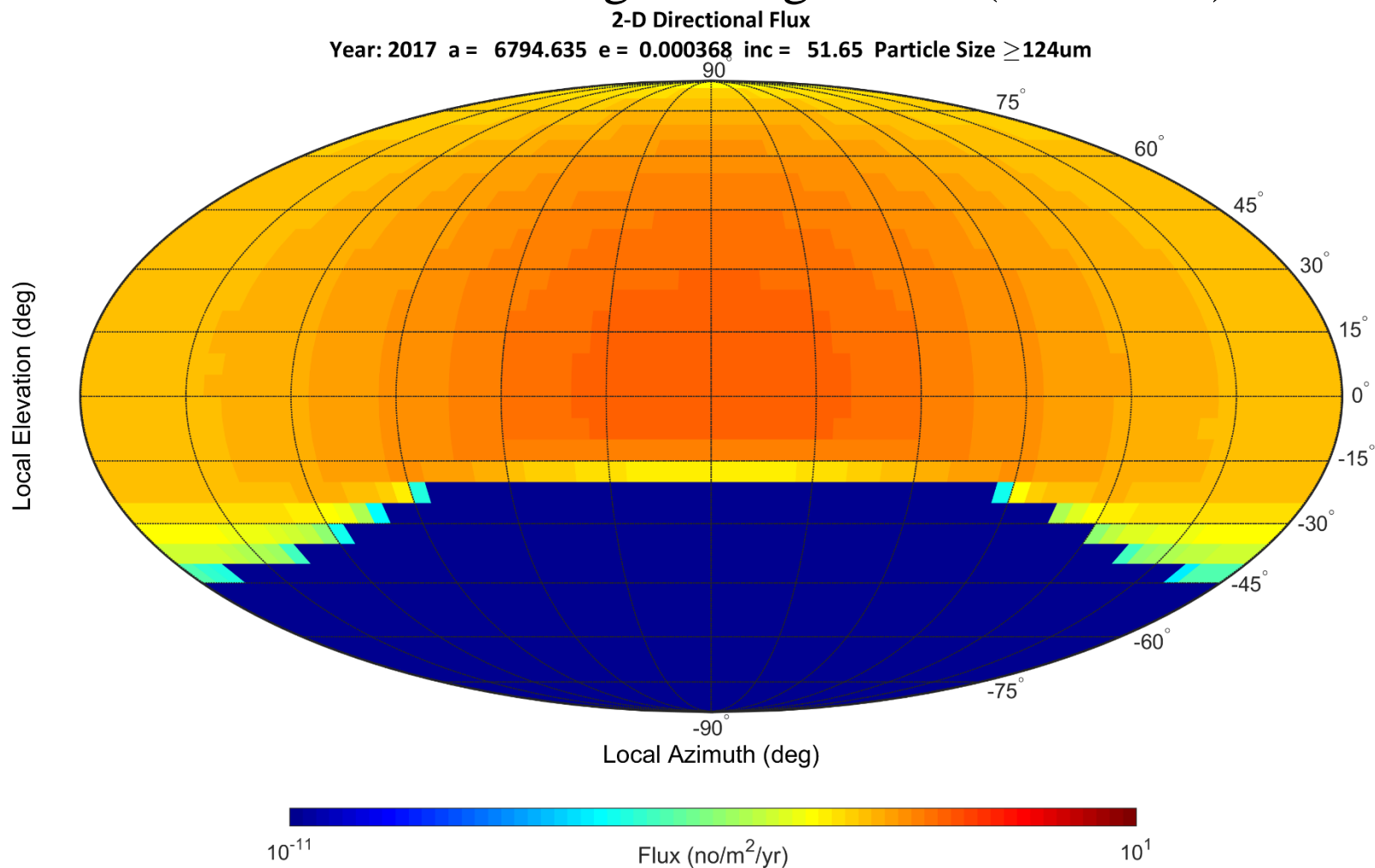
Orbital Debris Engineering Model (ORDEM 3.0)

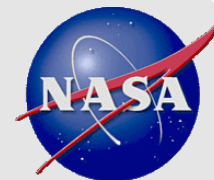




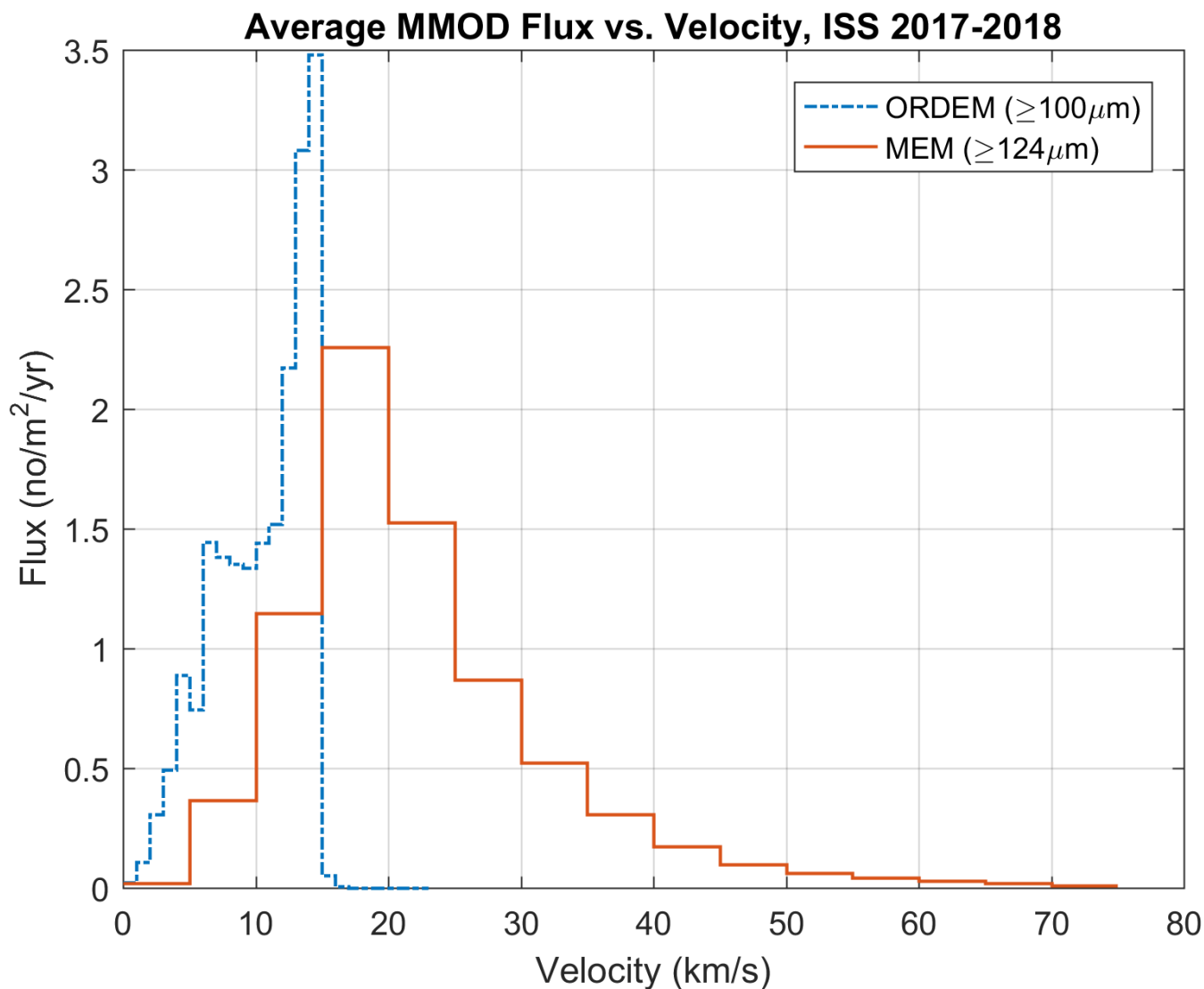
2-D Directional Flux – MEM 2.0

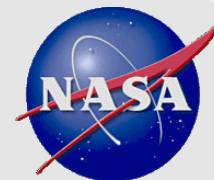
Micro-Meteoroid Engineering Model (MEM 2.0)





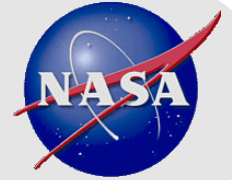
Predicted Flux vs. Velocity





Conclusions

- **SDS is the top priority for NASA ODPO development of orbital debris monitoring capability**
 - Addressing a gap in detection coverage
 - SDS will inform the design of future DRAGONS
- **The NASA ODPO will use the experience from SDS to improve the detection and characterization technology.**
 - Improved grids with 50μm width lines
 - Larger detection areas
 - Improved acoustic algorithms for speed, direction, and density calculations
- **The NASA ODPO is pursuing additional flight opportunities to put DRAGONS at higher altitudes**
 - Targeting flights in the 700 to 1000 km altitude region
 - Sun-synchronous orbits



Questions?

